

Amendments to the Claims:

The following listing of claims will replace all prior versions, and listings, of claims in the application:

1. (Original) A combustor for a gas turbine engine comprising a combustion chamber wall having formed therein at least one hole for admitting air into the combustion chamber;

at least one air intake chute aligned with said hole;

during operation a hoop stress field having regions of high and low stress concentration around said hole;

wherein said chute is attached to the combustor wall in a region of low stress concentration.
2. (Original) A combustor as claimed in claim 1 wherein the chute is attached to the combustor wall in at least two regions of low stress concentration.
3. (Original) A combustor as claimed in claim 2 wherein areas where the chute is attached to the combustor wall are substantially in the same radial plane.
4. (Original) A combustor as claimed in claim 2 wherein the areas of attachment are provided on diametrically opposite sides of said chute.
5. (Original) A combustor as claimed in claim 1 wherein the combustor air intake chute is provided with a flange disposed around one end thereof.
6. (Original) A combustor as claimed in claim 5 wherein the flange is circular.
7. (Original) A combustor as claimed in claim 5 wherein at least one tab projects from the outer edge of said flange.
8. (Original) A combustor as claimed in claim 7 wherein the at least one tab is attached to the combustor wall.

9. (Original) A combustor as claimed in claim 7 wherein the at least one tab projects from the edge of the flange up to about $0.14 \times$ (flange diameter).
10. (Original) A combustor as claimed in claim 7 wherein the at least one tab has a length of up to about $0.25 \times$ (flange diameter) of the diameter of the flange.
11. (Withdrawn) A method of manufacturing a combustor as described in claim 1 comprising the step of aligning the areas where the chute is attached to the combustor with the operational hoop stress field in the combustor wall.
12. (Withdrawn) A method of manufacturing a combustor as described in claim 8 comprising the step of aligning the areas where the chute is attached to the combustor with the operational hoop stress field in the combustor wall.
13. (Withdrawn) A method as claimed in claim 11 wherein the areas where the chute is attached to the combustor wall are orientated such that they are in the same radial plane.
14. (Withdrawn) A method as claimed in claim 12 wherein the areas where the chute is attached to the combustor wall are orientated such that they are in the same radial plane.
15. (Canceled)